

Orbits Optimisation Exercise

Description



EPSRC



NERC SCIENCE OF THE ENVIRONMENT



archer



CRAY
THE SUPERCOMPUTER COMPANY



epcc



Problem

- Imagine we want to launch a probe to land on a distant object
 - e.g. manned mission to Mars
- Need to reach a certain point at a certain time with a certain velocity
- In this exercise
 - given a fixed starting point, try to find the correct launch velocity that, after a fixed time, arrives at target point with correct velocity



Optimisation Problem

- Define an **objective function**
 - equals zero for correct solution
 - small (but positive) for a close solution
 - large (and positive) for a bad solution
- Run many simulations with different launch velocities
 - search for settings that minimise the objective function
- Definition of objective function: objf
 - actual position and velocity are (x,y) and (vx, vy)
 - target values are (xf, yf) and (vxf, vyf)
 - $\text{objf}^2 = ((x-x_f)/x_f)^2 + ((y-y_f)/y_f)^2 + ((v_x-v_{x_f})/v_{x_f})^2 + ((v_y-v_{y_f})/v_{y_f})^2$



Initial setup

Orbit simulation

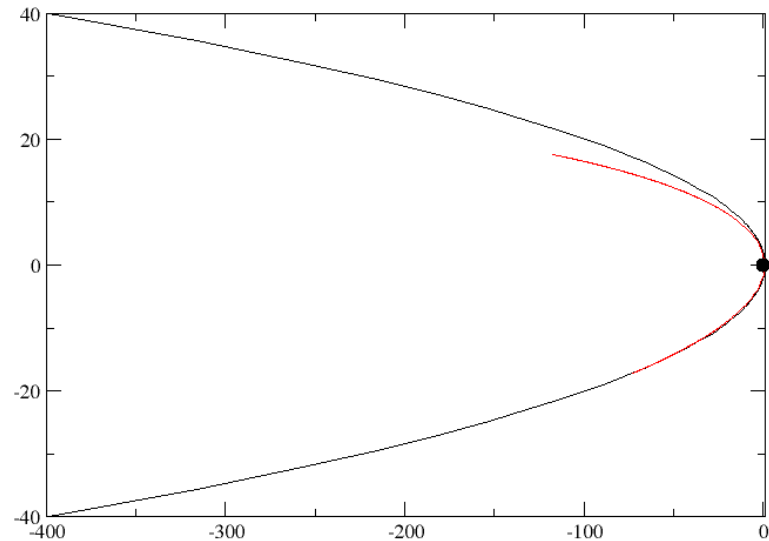
```
mass of object 1: 1000.000000
mass of object 2: 0.100000
end time for the integration: 30.000000
integration time step: 0.000100
```

Initial guess for velocity vector is (5.000000, 0.600000)

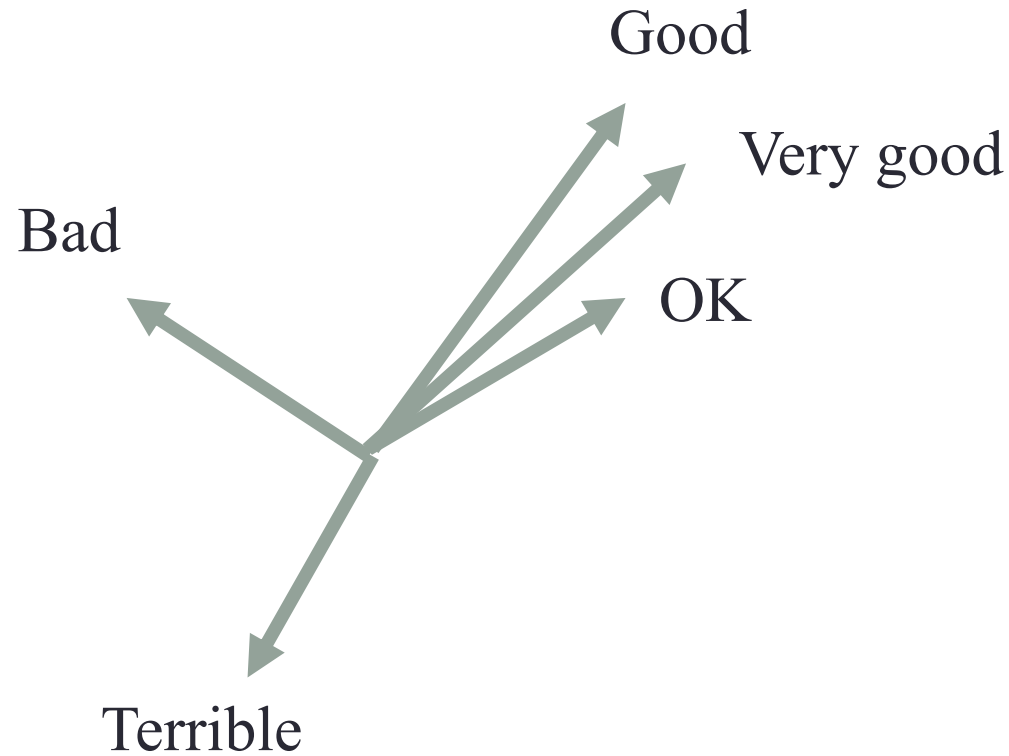
```
Starting time integration (300000 steps)...
... finished
```

```
Final position = (-117.489232, 17.561779), velocity = (-3.934065, 0.228667)
Target position = (-119.165926, 21.929982), velocity = (-4.046287, 0.369329)
```

Error in final values = 0.430924



Picking the velocity



Approaches

- Search points on a 2D grid that includes many velocities
 - pick the best
- Steepest descent
 - change the velocity and accept if it reduces the objective function
 - reduce the size of the change as we approach solution
- Feed into a black-box optimiser
 - “please minimise $f(v_x, v_y)$ ”
- etc. etc. ...

